

Gaussian Layer Connectivity Parameterization: A New Approach to Topology Optimization of Multi-Body Mechanisms

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Abstract

Topology optimization is extended to the design of multi-body mechanisms. A Gaussian function is used to parameterize the location of inter-body connections, thereby enabling the optimization of both mass distribution and inter-body connectivity simultaneously. The potential for large rigid-body rotation necessitates the use of a geometrically nonlinear finite element analysis to properly model mechanism response. The unknown displacement field is calculated with a Newton-Raphson iterative scheme. An adjoint analysis is performed to efficiently compute the design sensitivities used in the gradient-based optimization procedure. The proposed technique is demonstrated on the design of multi-body grippers and force/displacement inverters.

Keywords: topology optimization, multi-body mechanism design, nonlinear finite element analysis, adjoint sensitivity analysis

1. Introduction

2 Topology optimization applies nonlinear programming to algorithmically
3 design optimal structures [1]. Common design problems include compliance
4 minimization subject to a mass constraint [2] and mass minimization sub-
5 ject to a stress constraint [3]. As manufacturing technologies have advanced
6 in recent years, specifically additive manufacturing, the design process has
7 become more important to leverage the potential performance gains that
8 are now possible. For example, multi-scale design is nearly impossible to
9 optimize heuristically based on human intuition and experience. Computa-
10 tional design tools based on mathematical principles provide a much better
11 foundation for complex design problems [4].

12 Traditionally, a design domain is discretized into a finite number of cells,
13 each of which are assigned a design parameter. The optimal distribution
14 of material, or optimal structure, is then determined by solving the integer
15 programming problem of determining which cells should have material and
16 which cells should not [5]. To facilitate nonlinear programming, the problem
17 is relaxed by allowing continuously varying design parameters. The stiff-
18 ness of intermediate volume fractions, i.e. those parameters with a value
19 somewhere between solid and void, has been computed using a homoge-
20 nization technique [1, 6] or a material interpolation scheme [7, 8, 9]. Final
21 designs consisting only of design parameters corresponding to solid or void
22 can be obtained if the material interpolation scheme penalizes the stiffness-
23 to-weight ratio of intermediate design parameters and a resource constraint
24 is employed [5]. Numerical instabilities, such as “checkerboarding”, arise
25 from the numerical discretization [10], but may be alleviated with a geo-
26 metrically limiting constraint, such as a perimeter constraint [11], a slope
27 constraint [12], or a density filter [13].

28 The vast majority of mechanism design using topology optimization has
29 focused on the design of single-body structures [14], such as compliant mech-
30 anisms [15, 16]. However, multi-body structures have been the subject
31 of recent research [17, 18]. Numerical optimization has previously been
32 used to design the topology of flexible, multi-body mechanisms with fixed
33 connections [19] or to design the connectivity of rigid multi-body mecha-
34 nisms [20, 21, 22, 23]. This work proposes a method to simultaneously opti-
35 mize the topology and inter-body connectivity of flexible, planar multi-body
36 structures, which serve as an alternative to single-body compliant mecha-
37 nisms.

38 Optimizing element connectivity by allowing the stiffness of elastic links
39 connecting elements to vary is not a new approach. Single-body struc-
40 tures have been optimized by parameterizing the element connectivity rather
41 than the element volume fraction to avoid the numerical issues caused by
42 low volume fraction elements [24], and multi-body rigid block linkages have
43 been optimized by parameterizing inter-body elastic links to model revolute
44 joints [25, 26]. The proposed technique extends the idea of optimizing ele-
45 ment connectivity by parameterizing inter-body link elements to determine
46 optimal connectivity, while simultaneously parameterizing element volume
47 fractions to determine optimal topology. Previous work by the current au-
48 thors designated a design parameter to each inter-body link [27], while the
49 present work suggests a way to reduce the number of design parameters
50 utilizing a Gaussian parameterization for inter-body connectivity.

51 The ability to synthesize multi-body structures without prescribing the

52 inter-body connectivity a priori would enhance the effectiveness of design
53 algorithms and allow numerical optimization to improve the design process
54 for a new class of complex problems.

55 Multi-body mechanisms may undergo large motion as they are actu-
56 ated. To accurately model this motion, specifically rigid-body rotation, a
57 geometrically nonlinear finite element analysis is employed. A linear mate-
58 rial model is assumed despite use of the nonlinear Green-St. Venant strain,
59 since the majority of the displacement is expected to be from rigid body
60 rotation about pin joints rather than elastic distortion. The equilibrium
61 displacement field is computed using a Newton-Raphson iterative scheme.
62 A density filter allows local feature size control and avoids the “checker-
63 board” instability [10]. Optimal multi-body structures are obtained using
64 the method of moving asymptotes (MMA) [28], which requires design sen-
65 sitivities that are efficiently obtained using the adjoint method.

66 Section 2 provides the theoretical foundation of the proposed optimiza-
67 tion method. Section 3 demonstrates the method on the design on the
68 design of grippers and force/displacement inverters. Multi-body structures
69 with optimal mechanical and geometric advantage are explored.

70 **2. Theory and Methods**

71 Two-dimensional, planar structures are synthesized using a new topology
72 optimization technique. The designs are parameterized using two types
73 of design variables discussed in Section 2.1, modeled using a geometrically
74 nonlinear finite element analysis outlined in Section 2.3, and optimized using
75 the nonlinear programming methodology presented in Section 2.4.

76 *2.1. Design Parameterization*

77 Two-dimensional, planar mechanisms are modeled using multiple coinci-
78 dent finite element meshes. Each of these two-dimensional meshes is referred
79 to as a “layer”, and may contain zero, one, or more separate bodies. Each
80 layer is discretized with quadrilateral, continuum finite elements modeling
81 the elastic bodies and is connected to other layers by linear spring elements,
82 which are used to model pin joints. A visualization of the parameterization
83 of the planar mechanisms is displayed in Figure 1, for an example using 3
84 layers. In practice, the maximum number of layers must be prescribed be-
85 fore the optimization begins, however the optimizer may use only a fraction
86 of the allotted layers. Note that each mesh actually lies in the same plane,
87 although they are separated in a third dimension for the illustration.

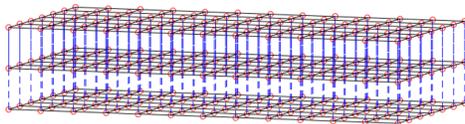


Figure 1: Visualization of element connectivity; \circ nodes, — body element edges, - - link elements

88 Two types of design parameters are used to describe each design, namely,
 89 body element stiffness parameters x_{body} and pin joint location parameters
 90 x_{link} . The volume fraction of each body element is assigned a x_{body} value
 91 that indicates the relative stiffness of that continuum element. A volume
 92 fraction of $x_{\text{body}} \approx 0$ corresponds to void or no material, while a volume
 93 fraction of $x_{\text{body}} = 1$ corresponds to solid material. A lower bound of $\rho_{\text{min}} =$
 94 1×10^{-3} is selected to avoid a singular consistent tangent operator that
 95 would result from a truly void element. Body element relative stiffness
 96 values are computed according to a SIMP interpolation scheme [7] given by

$$E = E_0 x_{\text{body}}^p, \quad (1)$$

97 where E is the effective Young's modulus of the element, E_0 is the Young's
 98 modulus of the base material, and p is a penalization exponent. This inter-
 99 polation scheme penalizes intermediate volume fractions by reducing their
 100 stiffness-to-weight ratio when $p > 1$, causing the optimizer to select values
 101 of ρ_{min} or 1 when a resource constraint is employed as they are the most
 102 economical. The derivative of a body element stiffness with respect to the
 103 corresponding volume fraction is given by

$$E = E_0 p x_{\text{body}}^{(p-1)}, \quad (2)$$

104 which will be needed for the sensitivity analysis performed in Section 2.3.4.

105 The bodies on separate layers interact through linear elastic springs, or
 106 “links”, connecting each pair of coincident nodes on prescribed layers. The
 107 stiffnesses of these “links” are distributed in such a way to model pin joints;
 108 the locations of which $(x_{\text{pin}}, y_{\text{pin}})$ are optimized. The stiffness of all link
 109 elements k_i is computed by evaluating a Gaussian stiffness distribution at
 110 the location of the link element (x_i, y_i) , which is given by

$$k_i(x_{\text{pin}}, y_{\text{pin}}) = K_0 \exp\left(-\left(\frac{(x_i - x_{\text{pin}})^2}{2\sigma_x^2} + \frac{(y_i - y_{\text{pin}})^2}{2\sigma_y^2}\right)\right), \quad (3)$$

111 where K_0 is the nominal stiffness of the pin joint and σ_x and σ_y are the
 112 standard deviations of the stiffness distribution in the x and y directions,

113 respectively. These link elements are assembled into the global finite element
114 system and serve to penalize relative displacement between the two nodes
115 that are connected. This technique aims to approximate the kinematics of
116 a pin joint, as proper selection of σ_x and σ_y will result in relatively high
117 stiffness at nodes near the candidate pin location (x_{pin}, y_{pin}) and relatively
118 low stiffness at surrounding nodes. There will be a nonzero stiffness at each
119 link in the domain, however, far away from the pin location the stiffness
120 is negligible due to the exponential nature of Equation (3). If there are
121 multiple pin joints connecting the same layers, their stiffness contributions
122 can be linearly superposed.

123 The derivatives of the inter-layer link stiffnesses with respect to the pin
124 coordinates are given by

$$\frac{\partial k_i(x_{pin}, y_{pin})}{\partial x_{pin}} = K_0 \exp\left(-\left(\frac{(x_i - x_{pin})^2}{2\sigma_x^2} + \frac{(y_i - y_{pin})^2}{2\sigma_y^2}\right)\right) \left(\frac{(x_i - x_{pin})}{\sigma_x^2}\right) \quad (4)$$

$$\frac{\partial k_i(x_{pin}, y_{pin})}{\partial y_{pin}} = K_0 \exp\left(-\left(\frac{(x_i - x_{pin})^2}{2\sigma_x^2} + \frac{(y_i - y_{pin})^2}{2\sigma_y^2}\right)\right) \left(\frac{(y_i - y_{pin})}{\sigma_y^2}\right), \quad (5)$$

126 which will be needed for the sensitivity analysis performed in Section 2.3.4.

127 2.2. Link Stiffness Estimation

128 The maximum link stiffness used to represent a pin joint was estimated
129 using a finite element analysis. A linear elastic model was used to calculate
130 the deflection of a three-dimensional cylinder under representative loading
131 in the commercial software ABAQUS. The bottom half of the pin was
132 fixed, while a distributed load of $P = 1 \text{ N m}^{-2}$ was applied to the top half
133 as displayed in Figure 2. The planar mechanisms discussed in Section 3
134 were assumed to have a thickness of $t = 10 \text{ mm}$ and overall size on the order
135 of $\approx 100 \text{ mm}$. Pins of length $l = 2t = 20 \text{ mm}$ with diameter varying from
136 $d = 1 \text{ mm} - 5 \text{ mm}$ were analyzed to give an estimate and range for realistic
137 pin stiffnesses K_0 . The stiffness was computed by dividing the total applied
138 load $F = \frac{Pdl}{2}$ by the displacement x of the midpoint of the top half of the
139 cylinder, which was estimated as half of the displacement of the top of the
140 pin. The results of the calculations are given in Table 1. A K_0 value of
141 $1 \times 10^5 \text{ N mm}^{-1}$ was selected for the optimizations presented in Section 3.

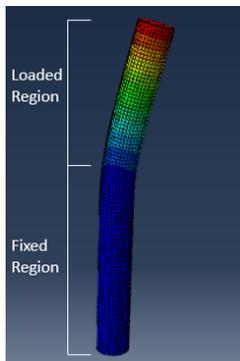


Figure 2: Finite element analysis used to estimate pin stiffness

Table 1: Computed Link Stiffnesses

Diameter (mm)	Length (mm)	F (N)	x (mm)	K_0 (N mm ⁻¹)
1	20	0.01	6.71×10^{-5}	1.49×10^2
2	20	0.02	8.65×10^{-6}	2.31×10^3
3	20	0.03	2.71×10^{-6}	1.11×10^4
4	20	0.04	1.23×10^{-6}	3.26×10^4
5	20	0.05	6.82×10^{-7}	7.34×10^4

142 *2.3. Geometrically Nonlinear Finite Element Analysis*

143 Multi-body mechanisms may undergo large motion upon actuation. A
 144 geometrically nonlinear finite element analysis is employed to model design
 145 responses, since large rotations are not accurately modeled using infinitesimal
 146 strain theory. Previous work has revealed that a nonlinear analysis can
 147 result in significant performance improvements over a linear analysis [29].
 148 This technique also has the advantage of capturing the load dependence of
 149 optimal designs, which has been investigated in previous nonlinear elastic
 150 optimizations [13]. The Green-St. Venant strain \mathbf{E} , defined by

$$\mathbf{E} = \frac{1}{2} (\mathbf{F}^T \mathbf{F} - \mathbf{I}), \quad (6)$$

151 where \mathbf{F} is the deformation gradient and \mathbf{I} is the identity tensor, does not
 152 produce strain through rigid-body rotations and will thus serve to accurately
 153 model large motion. The deformation gradient, which is the derivative of
 154 the deformed coordinates \mathbf{X} with respect to the original coordinates \mathbf{x} , can
 155 be equivalently expressed as $\mathbf{F} = \partial \mathbf{X} / \partial \mathbf{x}$ or $\mathbf{F} = \nabla \mathbf{U} + \mathbf{I}$, where \mathbf{U} is the

156 displacement field given by $\mathbf{U} = \mathbf{X} - \mathbf{x}$. The strong form of the equilibrium
 157 equations expressed over a general domain Ω , such as the one pictured in
 158 Figure 3, is given by

$$\nabla \cdot \mathbf{F}\mathbf{S} + \mathbf{b} = 0 \in \Omega \quad (7)$$

$$\mathbf{U} = \mathbf{U}^p \in A^u \quad (8)$$

$$\mathbf{F}\mathbf{S}\mathbf{n} = \mathbf{t}^p \in A^t, \quad (9)$$

161 where $\nabla \cdot$ is the divergence operator, \mathbf{S} is the Second Piola-Kirchoff stress, \mathbf{b}
 162 is the body force per unit volume, \mathbf{U}^p are the prescribed displacements which
 163 are applied to the fixed boundary region A^u , \mathbf{n} is the unit normal vector,
 164 and \mathbf{t}^p are the prescribed tractions which are applied to the boundary region
 165 A^t . Equations (7)-(9) are expressed in an equivalent weak form given by

$$\int_{\Omega} \mathbf{w} \cdot (\nabla \cdot \mathbf{F}\mathbf{S} + \mathbf{b}) \, d\Omega + \int_{A^t} \mathbf{w} \cdot (\mathbf{F}\mathbf{S}\mathbf{n} - \mathbf{t}^p) \, dA^t = 0, \quad (10)$$

166 where \mathbf{w} are weighting functions. Integration by parts and the divergence
 167 theorem are applied to recast Equation (10) into a form suitable for dis-
 168 cretization by the Galerkin finite element method [30].

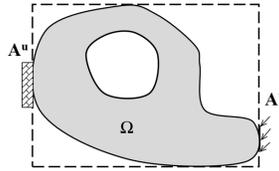


Figure 3: Boundary value problem domain

169 A linear elastic material model is assumed, with a constitutive relation
 170 given by

$$\mathbf{S} = \mathbf{C}\mathbf{E}, \quad (11)$$

171 where \mathbf{C} is the constitutive tensor which is independent of displacement.
 172 This model is motivated by the expectation that multi-body mechanisms
 173 may undergo large displacements, yet small strains. The inclusion of pin
 174 joints will allow large relative rotations to take place between bodies without
 175 large material distortions.

176 2.3.1. Newton-Raphson Solution Procedure

177 The iterative Newton-Raphson method is used to calculate equilibrium
 178 displacement solutions \mathbf{U} that satisfy

$$\mathbf{R}(\mathbf{U}) = \mathbf{F}_{\text{ext}} - \mathbf{F}_{\text{int}}(\mathbf{U}) = 0, \quad (12)$$

179 where $\mathbf{R}(\mathbf{U})$ is the residual vector that is to be minimized. The external
 180 force vector, \mathbf{F}_{ext} , and internal force vector, $\mathbf{F}_{\text{int}}(\mathbf{U})$, are deduced from Equa-
 181 tion (10). A first-order Taylor series approximation is used to compute the
 182 displacement update $\Delta\mathbf{U}$ at each iteration i that causes the residual to be
 183 zero, or $\mathbf{R}(\mathbf{U}_i + \Delta\mathbf{U}) = 0$. The displacement update is calculated according
 184 to

$$\mathbf{K}(\mathbf{U}_i) \Delta\mathbf{U} = \mathbf{R}(\mathbf{U}_i), \quad (13)$$

185 where \mathbf{K} is the consistent tangent operator defined as $\mathbf{K}(\mathbf{U}_i) = -\frac{\partial\mathbf{F}_{\text{int}}(\mathbf{U}_i)}{\partial\mathbf{U}_i}$.
 186 The displacement solution is updated each iteration by $\mathbf{U}_{i+1} = \mathbf{U}_i + \Delta\mathbf{U}$.
 187 This iterative process continues until Equation (12) converges to within a
 188 satisfactory tolerance.

189 2.3.2. Displacement Damping

190 Dirichlet boundary conditions are necessary to obtain a non-singular
 191 consistent tangent operator and find a unique solution to the unknown dis-
 192 placement field. Proper selection of boundary conditions proved to be a
 193 challenge in the multi-body framework as there were multiple meshes to
 194 constrain and the function of each layer was not known a priori. Ade-
 195 quate boundary conditions often overly restricted the design, and so “anchor
 196 springs” were attached to act as displacement dampers. All nodes in each
 197 layer were attached to a fixed node (i.e. a node with a Dirichlet boundary
 198 condition of zero displacement) by linear spring elements with relatively low
 199 stiffness. With sufficient “anchor spring” stiffness, or displacement damping,
 200 the solution could converge under Dirichlet boundary conditions that would
 201 otherwise contain rigid body displacement modes.

202 It was often observed in practice that the pin joints between layers re-
 203 moved the rigid body displacement modes, thus adequately constraining the
 204 multi-body mesh. However, the optimization was prone to instability and
 205 divergence if regions of the design space exist where the mesh was not con-
 206 strained properly. The inclusion of displacement damping resolved this issue
 207 and allowed the optimization to progress. “Anchor spring” stiffnesses are
 208 provided in Section 3.

209 2.3.3. Density Filtering

210 For a given volume fraction, a structure with more smaller holes is stiffer
 211 than a structure with fewer larger holes [10]. An ill-posed optimization prob-
 212 lem therefore results from a finite element discretization, as optimal designs
 213 are a function of the mesh resolution; the optimizer will create smaller fea-
 214 tures as the mesh is refined. This “checkerboard” instability can be rectified

215 by including a geometrically limiting constraint. A density filter that penal-
 216 izes designs with features smaller than the filter radius is employed to create
 217 a well-posed optimization problem.

218 Filtered volume fractions x_{body} , which are used in the finite element
 219 analysis, are computed as a function of unfiltered volume fractions \bar{x}_{body} ,
 220 which are the independent design variables. Their relation is given in matrix
 221 form by

$$\mathbf{x}_{\text{body}} = \mathbf{W}\bar{\mathbf{x}}_{\text{body}}. \quad (14)$$

222 where \mathbf{W} is the filtering matrix [13]. The components of \mathbf{W} are determined
 223 by first computing $\tilde{\mathbf{W}}$ according to

$$\tilde{W}_{i,j} = \max(0, \varepsilon - d_{i,j}), \quad (15)$$

224 where ε is the filter radius and $d_{i,j}$ is the distance between elements i and
 225 j . Each row of $\tilde{\mathbf{W}}$ is normalized to obtain \mathbf{W} , ensuring that filtered volume
 226 fractions are in the range of 0 and 1.

227 2.3.4. Adjoint Sensitivity Analysis

228 An adjoint analysis is used to compute the sensitivities of response func-
 229 tions with respect to the design parameters. The adjoint expressions are
 230 derived as follows. Consider a general response function of the form

$$f = \Pi(\mathbf{U}(\mathbf{x}), \mathbf{P}(\mathbf{x}), \mathbf{x}), \quad (16)$$

231 where \mathbf{U} is the vector of all nodal displacements, \mathbf{P} is the vector of all
 232 nodal forces, and \mathbf{x} is the vector of design parameters. The derivative of
 233 Equation (16) for response functions containing unknown displacements \mathbf{U}^f
 234 and unknown reaction forces \mathbf{P}^p will contain implicit derivatives that are
 235 expensive to compute. The superscripts f and p denote the partitions of
 236 the finite element vectors which correspond to degrees of freedom with free
 237 and prescribed displacements, respectively. The adjoint method permits the
 238 removal of these implicit terms from the sensitivity expression and is com-
 239 putationally advantageous over a direct method for optimization problems
 240 containing more design variables than constraint functions. Equation 16 can
 241 be re-written as

$$f = \Pi(\mathbf{U}(\mathbf{x}), \mathbf{P}(\mathbf{x}), \mathbf{x}) + \boldsymbol{\lambda}^T \mathbf{R}, \quad (17)$$

242 where \mathbf{R} is the residual vector and $\boldsymbol{\lambda}$ is an arbitrary constant vector, since
 243 the addition of zero will not change the right hand side and the inner prod-
 244 uct of any vector with the zero vector will remain zero. The accuracy of

245 the sensitivities are therefore reliant upon the tolerance of the linear solu-
 246 tion obtained from the Newton-Raphson procedure, as the residual vector
 247 is assumed here to be zero. Differentiating Equation (17) with respect to
 248 a particular design parameter under the assumption of design independent
 249 loading gives a sensitivity expression via the chain rule of

$$\frac{d\Pi}{dx} = \frac{\partial\Pi}{\partial\mathbf{U}^f} \frac{d\mathbf{U}^f}{dx} + \frac{\partial\Pi}{\partial\mathbf{P}^p} \frac{d\mathbf{P}^p}{dx} + \frac{\partial\Pi}{\partial x} + \boldsymbol{\lambda}^{fT} \frac{d\mathbf{R}^f}{dx} + \boldsymbol{\lambda}^{pT} \frac{d\mathbf{R}^p}{dx}. \quad (18)$$

250 Derivatives with respect to body element parameters refer to filtered volume
 251 fractions, although derivative with respect to unfiltered volume fractions can
 252 be easily obtained by differentiating Equation (14) and employing the chain
 253 rule.

254 Two types of functions will be considered here and will serve to form a
 255 basis for all objective and constraint functions. The first response function
 256 is of the form

$$f_1 = \mathbf{L}^{fT} \mathbf{U}^f, \quad (19)$$

257 where \mathbf{L}^f is a constant vector used to define a response function involving
 258 unknown displacements. Selecting the prescribed and free partitions of the
 259 arbitrary vector, $\boldsymbol{\lambda}$, such that

$$\boldsymbol{\lambda}^p = \mathbf{0} \quad \boldsymbol{\lambda}^f = \mathbf{K}^{ff^{-1}} \mathbf{L}^f, \quad (20)$$

260 reduces the sensitivity expression to

$$\frac{d\Pi}{dx} = -\boldsymbol{\lambda}^{fT} \frac{\partial\mathbf{F}_{\text{int}}^f}{\partial x} = -\left(\mathbf{K}^{ff^{-1}} \mathbf{L}^f\right)^T \frac{\partial\mathbf{F}_{\text{int}}^f}{\partial x}. \quad (21)$$

261 Note that when $\mathbf{L}^f = \mathbf{P}^f$, i.e. the response function is the compliance, there
 262 is no need to solve a new linear system for the adjoint vector as $\boldsymbol{\lambda}^f = \mathbf{U}^f$.

263 The second response function is of the form

$$f_2 = \mathbf{L}^{pT} \mathbf{P}^p, \quad (22)$$

264 where \mathbf{L}^p is a vector used to define a response function involving reaction
 265 forces. Selecting the prescribed and free partitions of the arbitrary vector,
 266 $\boldsymbol{\lambda}$, such that

$$\boldsymbol{\lambda}^p = -\mathbf{L}^p \quad \boldsymbol{\lambda}^f = -\mathbf{K}^{ff^{-1}} \left(\mathbf{K}^{fp} \mathbf{L}^p\right), \quad (23)$$

267 reduces the sensitivity expression to

$$\frac{d\Pi}{dx} = -\boldsymbol{\lambda}^{fT} \frac{\partial\mathbf{F}_{\text{int}}^f}{\partial x} - \boldsymbol{\lambda}^{pT} \frac{\partial\mathbf{F}_{\text{int}}^p}{\partial x} = -\left(\mathbf{K}^{ff^{-1}} \left(\mathbf{K}^{fp} \mathbf{L}^p\right)\right)^T \frac{\partial\mathbf{F}_{\text{int}}^f}{\partial x} - (\mathbf{L}^p)^T \frac{\partial\mathbf{F}_{\text{int}}^p}{\partial x}. \quad (24)$$

268 By selecting the above-defined adjoint solution for each function of in-
 269 terest, we cause all terms containing implicit derivatives to vanish from the
 270 sensitivity expression, thereby increasing the computational efficiency of the
 271 sensitivity calculation. The explicit derivatives of the internal force vector
 272 with respect to the design parameters, $\frac{\partial \mathbf{F}_{\text{int}}^f}{\partial \mathbf{x}}$, can be obtained with the help
 273 of Equations (2), (4), and (5).

274 2.4. Optimization Problem

275 Nonlinear programming is used to solve optimization problems of the
 276 form

$$\begin{aligned}
 & \underset{\mathbf{x}}{\text{minimize}} && f(\mathbf{x}) \\
 & \text{subject to} && g_i(\mathbf{x}) \leq 0, && i = 1, 2, \dots, N_{\text{constraint}} \\
 & && x_{\min j} \leq x_j \leq x_{\max j}, && j = 1, 2, \dots, N_{\text{param}}
 \end{aligned} \tag{25}$$

277 where \mathbf{x} is the vector of design parameters, $f(\mathbf{x})$ is the objective function,
 278 $g_i(\mathbf{x})$ are the constraint functions, and $x_{\min j}$ and $x_{\max j}$ denote the lower and
 279 upper bounds for each design parameter x_j . All numerical optimizations in
 280 this work were solved using the method of moving asymptotes (MMA) [28].
 281 The inclusion of step limits for the pin location parameters was necessary to
 282 reduce the norm of the KKT conditions to a suitable tolerance. This will
 283 be discussed in further detail in Section 3.

284 Two design problems will be addressed. The first is that of mechanical
 285 advantage, or the ratio of output force F_{out} to input force F_{in} . The formal
 286 optimization problem is given by

$$\begin{aligned}
 & \underset{\mathbf{x}}{\text{minimize}} && \frac{-F_{\text{out}}}{F_{\text{in}}} = \frac{\mathbf{L}^p \mathbf{T} \mathbf{P}^p}{\mathbf{L}^f \mathbf{T} \mathbf{P}^f} \\
 & \text{subject to} && \mathbf{R}(\mathbf{U}) = 0 \\
 & && \mathbf{U} = \mathbf{U}^p \in A^u \\
 & && \mathbf{F} \mathbf{S} \mathbf{n} = \mathbf{t}^p \in A^t \\
 & && \mathbf{C} = \mathbf{P}^f \mathbf{T} \mathbf{U}^f \leq \mathbf{C}_{\text{max}} \\
 & && \frac{1}{N_{\text{body } k}} \sum_{j=1}^{N_{\text{body } k}} x_{\text{body } j} \leq V_{\text{max } k} \\
 & && \rho_{\min} \leq x_{\text{body } m} \leq 1 && m = 1, 2, \dots, N_{\text{body}} \\
 & && z_{\min} \leq x_{\text{link } n} \leq z_{\max} && n = 1, 2, \dots, N_{\text{link}},
 \end{aligned} \tag{26}$$

287 where \mathbf{L}^p is a vector defined here to select desired reaction forces, \mathbf{L}^f is a
 288 vector defined here to select desired input forces, C is the compliance of the
 289 structure, N_{body_k} is the number of body elements on a given layer k , $V_{\text{max}k}$
 290 is the maximum volume fraction of layer k , and z_{min} and z_{max} are domain
 291 specific bounds on pin location. A maximum compliance C_{max} constraint is
 292 necessary to ensure that the structure has a requisite level of stiffness.

293 The second design problem studied is that of geometric advantage, or
 294 the ratio of output displacement U_{out} to input displacement U_{in} . The formal
 295 optimization problem is given by

$$\begin{aligned}
 & \underset{\mathbf{x}}{\text{minimize}} && \frac{-U_{\text{out}}}{U_{\text{in}}} = \frac{\mathbf{L}^f \mathbf{T} \mathbf{U}^f}{\mathbf{L}^p \mathbf{T} \mathbf{U}^p} \\
 & \text{subject to} && \mathbf{R}(\mathbf{U}) = 0 \\
 & && \mathbf{U} = \mathbf{U}^p \in A^u \\
 & && \mathbf{F} \mathbf{S} \mathbf{n} = \mathbf{t}^p \in A^t \\
 & && F_{\text{in}} = \mathbf{L}^p \mathbf{T} \mathbf{P}^p \geq F_{\text{min}} \\
 & && \frac{1}{N_{\text{body}_k}} \sum_{j=1}^{N_{\text{body}_k}} x_{\text{body}_j} \leq V_{\text{max}k} \\
 & && \rho_{\text{min}} \leq x_{\text{body}_m} \leq 1 \quad m = 1, 2, \dots, N_{\text{body}} \\
 & && z_{\text{min}} \leq x_{\text{link}_n} \leq z_{\text{max}} \quad n = 1, 2, \dots, N_{\text{link}}
 \end{aligned} \tag{27}$$

296 where \mathbf{L}^p is a vector defined here to select desired input displacements, \mathbf{L}^f
 297 is a vector defined here to select desired output displacements. For certain
 298 boundary conditions a minimum force F_{min} may need to be enforced to
 299 achieve reasonable designs. A maximum force may be imposed in place of
 300 or in addition to a minimum force if needed for the application.

301 3. Results and Discussion

302 Two common topology optimization design problems are explored in a
 303 multi-body framework. First, a gripper is synthesized using two layers to
 304 display the method's ability to optimize material distribution and inter-body
 305 connectivity simultaneously. The optimal designs for mechanical advantage
 306 and geometric advantage are computed for a range of input parameters.
 307 The second design problem investigated is that of a force or displacement
 308 inverter. Optimal designs are computed for mechanical advantage, corre-
 309 sponding to a force inverter, and geometric advantage, corresponding to

310 a displacement inverter, on a design domain consisting of 3 body element
 311 layers.

312 3.1. Multi-Body Gripper Design Conditions

313 A common example problem in traditional, single-body topology opti-
 314 mization is that of a compliant gripper mechanism. The basic function is
 315 to magnify input displacements or forces to gain a geometric or mechanical
 316 advantage, respectively. An example compliant gripper optimized for me-
 317 chanical advantage is presented in Figure 4. This mechanism relies on elastic
 318 distortion for actuation and would not function at all if the material were
 319 perfectly rigid. Despite the simplicity of a single part, this design has the dis-
 320 advantage that energy is lost to elastic distortion and it may be susceptible
 321 to fatigue due to the flexural hinges that create stress concentrations [16].

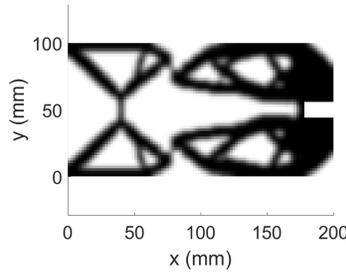


Figure 4: Example compliant gripper design

322 The proposed method aims to design a gripper that utilizes a pin joint,
 323 rather than elastic distortion, to increase mechanical efficiency and reduce
 324 material fatigue. The multi-body design domain consists of two body ele-
 325 ment layers connected by one link element layer with one Gaussian modeled
 326 pin joint, which is initialized to $(x, y) = (90, 50)$. In all optimizations, an
 327 initial volume fraction equal to the maximum volume fraction constraint is
 328 applied to each element on each layer. The gripper is actuated with equal
 329 and opposite vertical forces or displacements on the left side of the domain,
 330 while output forces or displacements are desired at a prescribed location on
 331 the left side of the domain, as pictured in Figure 5.

332 Symmetry was imposed between the two layers by associating the vol-
 333 ume fraction of mirrored elements in opposing layers with the same design
 334 parameter. For example, the volume fraction of the element in the top left
 335 corner of layer 1 corresponded to the same design parameter as the vol-
 336 ume fraction of the element in the bottom left corner of layer 2. Enforcing

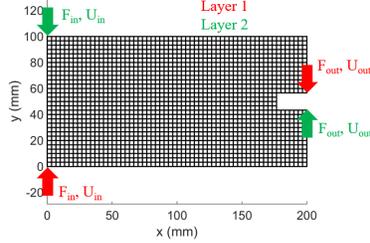


Figure 5: Domain and boundary conditions for the gripper design problem

337 symmetry reduced the dimensionality of the numerical optimization problem
 338 therefore reducing computational cost, and ensured the grippers were
 339 ambidextrous, which is advantageous for manufacturing and usage.

340 The parameters used in the multi-body optimizations are presented in
 341 Table 2. Parameters specific to the gripper design problem are presented
 342 in Table 3. Note that the input displacement only applies to the geometric
 343 advantage optimization, and the input force only applies to the mechanical
 344 advantage optimization. Additionally, the output spring stiffness has no
 345 bearing on mechanical advantage results as the output node is fixed. A
 346 maximum volume fraction is imposed on each layer, and the reported total
 347 maximum volume fraction in Table 3 is a sum of all layer maximum volume
 348 fractions.

349 Input displacements are applied through Dirichlet boundary conditions,
 350 and input forces are applied through Neumann boundary conditions. Out-
 351 put displacements were obtained by computing the displacement at desired
 352 nodes, although it should be noted that in practice a spring was connected
 353 between output nodes and a fixed node to require material be present at
 354 the output location. The displacement of a node attached to a spring is
 355 analogous to output force in a linear finite element analysis, however, the
 356 use of a spring with relatively low stiffness and a geometrically nonlinear
 357 finite element analysis justify this definition of output displacement in the
 358 current work. Output forces were obtained by fixing the desired nodes with
 359 Dirichlet boundary conditions and computing the associated reaction forces.

360 The use of a Gaussian distribution of stiffness to model a pin joint will
 361 affect the accuracy of the design evaluations. The non-zero stiffness at neigh-
 362 boring nodes, which allowed the pin joint to migrate throughout the domain
 363 without falling into local minima, will not behave the same as a pin joint
 364 with zero rotational stiffness. The standard deviations were set to the same
 365 length as the node spacing, or $\sigma_x = \sigma_y = \Delta_x = \frac{L_x}{(N_x-1)}$. If the Gaussian

Table 2: Multi-Body Mechanism Optimization Conditions

Description	Symbol	Value
Sum of Anchor Spring Stiffnesses	k_{anch}	$1 \times 10^2 \text{ N mm}^{-1}$
Output Spring Stiffnesses	k_{out}	$1 \times 10^4 \text{ N mm}^{-1}$
Maximum Link Stiffness	K_0	$1 \times 10^5 \text{ N mm}^{-1}$
Link Standard Deviation	σ_x, σ_y	2 mm
Young's Modulus	E_0	200 GPa
Poisson's Ratio	ν	0.3
Minimum Volume Fraction	x_{min}	1×10^{-3}
Maximum Volume Fraction	x_{max}	1.0
SIMP Penalty Exponent	p	3
Density Filter Radius	ε	5 mm
Domain Thickness	t	10 mm
Input Force Magnitude	F_{in}	$1 \times 10^2 \text{ N}$
Input Displacement Magnitude	U_{in}	$1 \times 10^{-2} \text{ mm}$

Table 3: Gripper Design Specific Conditions

Description	Symbol	Value
Domain Length	L_x	200 mm
Domain Height	L_y	100 mm
Number of Nodes in x-direction	N_x	101
Number of Nodes in y-direction	N_y	51
Maximum Total Volume Fraction	$V_{\text{total,max}}$	0.75
Maximum Compliance	C_{max}	2 N mm

366 stiffness distribution was located exactly at a particular node, the closest
 367 node (one grid spacing away) would have 61% of the maximum stiffness.
 368 The performance accuracy will be estimated by comparing the calculated
 369 mechanical advantage of grippers with the theoretical mechanical advantage
 370 that comes from the statics problem based only on pin joint location, i.e.
 371 equating moments caused by input and output forces.

372 3.2. Multi-Body Gripper Numerical Results

373 The gripper design problems aim to determine both the optimal distribu-
 374 tion of material along with the optimal pin coordinates. A design optimizing
 375 geometric advantage is presented in Figure 6. This design is not typical of
 376 everyday pliers, as the optimal pin joint location, $(x_0^*, y_0^*) = (32.5, 50.0)$,
 377 is much closer to the input locations than the output locations. However,

378 it is intuitive to see that small input displacements would be magnified by
 379 this pin joint location, thus creating large output displacements. This de-
 380 sign achieved a geometric advantage of 1.51, further justifying the technique
 381 used to obtain output displacements discussed in Section 3.1. An optimal
 382 pin joint location much closer to the output locations would be expected if
 383 the inclusion of springs attached to output nodes were erroneously modeling
 384 output force. The design in Figure 6 stored 10.0 N mm of strain energy un-
 385 der the loading conditions used for the optimization, which was a significant
 386 reduction from 14.3 N mm of stored strain energy in a single-body compliant
 387 mechanism optimized under the same parameters.

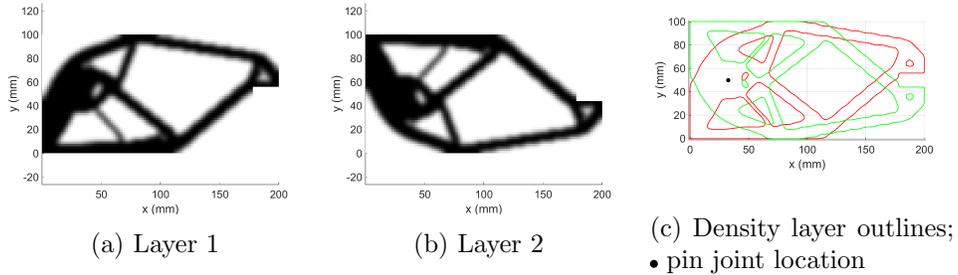


Figure 6: Optimal geometric advantage two-layer gripper

388 Optimizing mechanical advantage required an additional response con-
 389 straint. As mentioned in Section 2.4, a compliance constraint is necessary to
 390 ensure the structure has a requisite level of stiffness. Absent this constraint
 391 the optimizer will almost certainly remove material from the input location,
 392 and the resulting singular tangent operator will cause the optimization to
 393 fail. Grippers with optimal mechanical advantage subject to a compliance
 394 constraint are presented in Figure 7. As expected, the optimal pin joint
 395 location, $(x_0^*, y_0^*) = (139.6, 50.0)$, is now closer to the output location
 396 than the input location. If the compliance constraint is relaxed x_0^* will increase,
 397 and if the compliance constraint is tightened x_0^* will decrease. If we as-
 398 sume that the body is perfectly rigid, the theoretical mechanical advantage
 399 can be computed by equating the moments caused by the input and out-
 400 put forces, or $F_{in}(x_0^*) = F_{out}(L_x - x_0^*)$. Thus, the mechanical advantage is
 401 computed as $F_{out}/F_{in} = x_0^*/(L_x - x_0^*)$, which gives a value of 2.31 for the
 402 design in Figure 7. The optimal mechanical advantage computed with the
 403 Gaussian stiffness pin model was 2.26, or about 2% different than the the-
 404 oretical response. The difference between the theoretical response and the
 405 computed response is due the fact that a portion of the work imparted to

406 the mechanism by the user, is lost as stored elastic strain energy. The design
 407 in Figure 7 stored 1.81 N mm of strain energy under the loading conditions
 408 used for the optimization, which was a reduction from 2.00 N mm of stored
 409 strain energy in a single-body optimized compliant mechanism optimized
 410 under the same parameters.

411 Iteration histories are provided for the grippers optimized for mechanical
 412 advantage. The density fields are displayed in Figure 8, while the mechanical
 413 advantage and KKT norm are displayed in Figure 9. The KKT norm was
 414 reduced 3 orders of magnitude in 38 MMA iterations, during which the
 415 mechanical advantage increased from less than 1.0 in the initial configuration
 416 to its final value of 2.26.

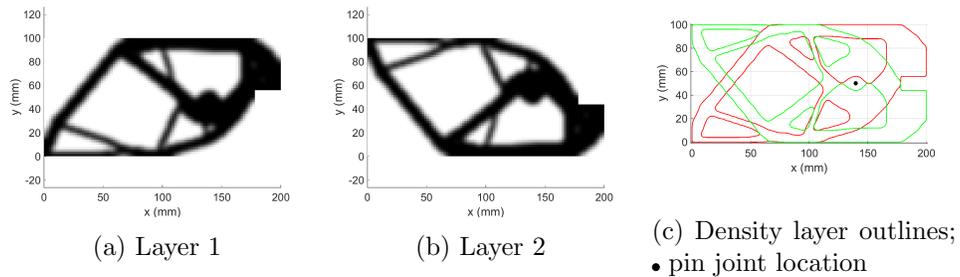


Figure 7: Optimal mechanical advantage two-layer gripper

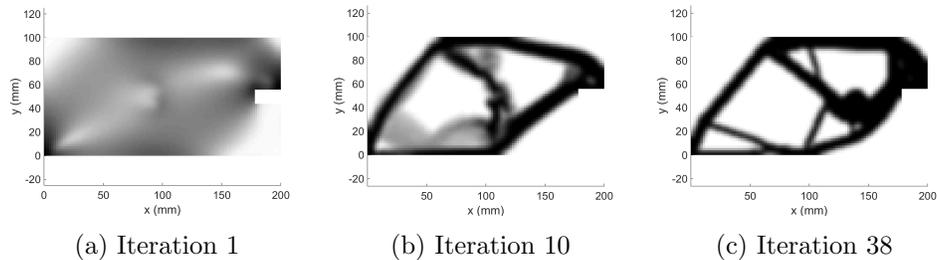


Figure 8: Iteration history of layer 1 density field

417 The mechanical advantage design problem has inherently small displace-
 418 ments. Fixing the output location is equivalent to putting a very stiff object
 419 in the grips of a set of pliers, which restricts all motion except for elastic
 420 distortion. For relatively small input forces, or those assumed to be applied
 421 by human hands on a steel set of pliers, there will be very small displace-
 422 ments. The geometric advantage design problem, however, lends itself to

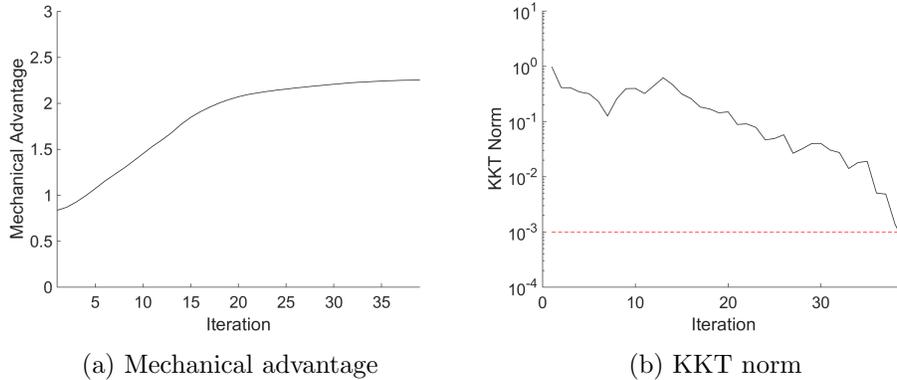


Figure 9: Iteration histories

423 much larger displacements. The output location is allowed to move, which
 424 means rigid body rotations are admissible. Despite this, the applied input
 425 displacement was only 1×10^{-1} mm, or 0.1% of the domain height. This
 426 small value was selected for numerical reasons. The inclusion of void ele-
 427 ments in the finite element analysis can result in poor behavior when these
 428 low volume fraction elements are distorted. This problem has been addressed
 429 previously by removing void elements [31], however a simpler technique of
 430 applying small input displacements to avoid the numerical instabilities was
 431 employed here.

432 Solutions were considered optimal once the Karush-Kuhn-Tucker (KKT)
 433 conditions [32] had converged to a tolerance of 1×10^{-3} , which was about
 434 3 orders of magnitude lower than the initial KKT norm. As previously
 435 mentioned, step limits were imposed on the pin joint location parameters.
 436 Specifically, maximum step sizes per iteration of 2% of the parameter range
 437 were enforced to prevent oscillation of the pin location and promote stable
 438 convergence.

439 3.3. Force/Displacement Inverter Design Conditions

440 A force or displacement inverter is another common example problem in
 441 single-body topology optimization. The basic function of the mechanism is
 442 to change the direction of input forces or displacements by 180° . An exam-
 443 ple single-body compliant design which maximized mechanical advantage is
 444 displayed in Figure 10. As with the gripper, the proposed method aims to
 445 design an inverter that utilizes multiple bodies and pin joints to increase
 446 efficiency and reduce material fatigue.

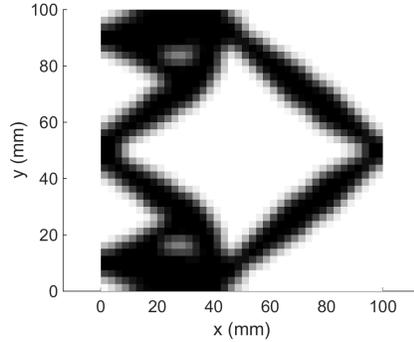


Figure 10: Example compliant force inverter design

447 Design problems will be solved on computational domains consisting of
 448 three body element layers, with two layers of link elements (connecting layers
 449 1 to 2 and layers 2 to 3). The locations of the two pins connecting layers
 450 1 to 2 are initialized to $(x, y) = (40, 40)$ and $(40, 60)$, while the locations of
 451 the two pin pins connecting layers 2 to 3 are initialized to $(x, y) = (60, 40)$
 452 and $(60, 60)$. No symmetries were enforced on this domain. The inverter
 453 is actuated by a horizontal input force or displacement on the left side of
 454 the domain, while a horizontal output force or displacement in the opposite
 455 direction is desired at a prescribed location on the left side of the domain,
 456 as pictured in Figure 11.

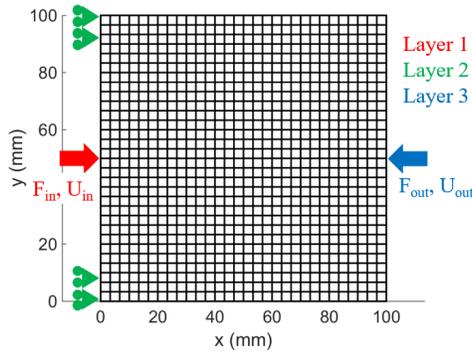


Figure 11: Domain and boundary conditions for the inverter design problem

457 The parameters specific to multi-body inverter optimizations are listed
 458 in Table 4. The parameters listed in Table 2 are also applicable. Again,
 459 note that input displacement only applies to the geometric advantage prob-
 460 lem, input force only applies to the mechanical advantage problem, and the

Table 4: Displacement/Force Inverter Design Specific Conditions

Description	Symbol	Value
Domain Length	L_x	100 mm
Domain Height	L_y	100 mm
Number of Nodes in x-direction	N_x	51
Number of Nodes in y-direction	N_y	51
Maximum Volume Fraction	V_{\max}	0.3
Maximum Compliance	C_{\max}	0.5 N mm

461 output spring stiffness has no bearing on mechanical advantage results. The
 462 same techniques for input and output displacement and force discussed in
 463 Section 3.1 are employed.

464 *3.3.1. Force/Displacement Inverter Numerical Results*

465 Optimal displacement inverters are pictured in Figure 12, and optimal
 466 force inverters are pictured in Figure 13. An encouraging result seen in
 467 both Figures 12 and 13 is that multiple distinct bodies can form on the
 468 same layer. Specifically, layer 2 in both designs contains two bodies near
 469 the fixed boundary conditions. This capability could greatly reduce the
 470 dimensionality of multi-body mechanisms by reducing the required number
 471 of layers needed to represent them.

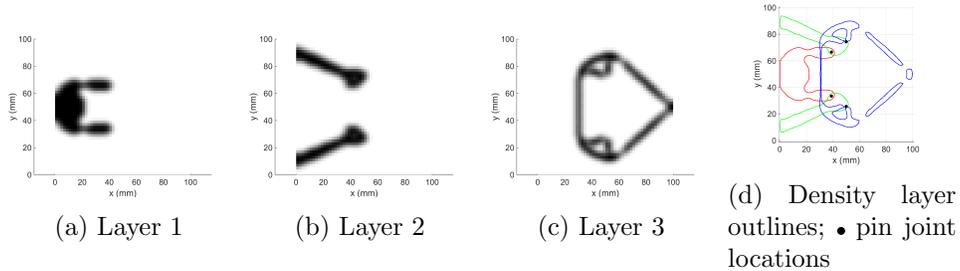


Figure 12: Optimal three-body displacement inverter

472 The current designs rely on a combination of elastic distortion and rigid
 473 body rotation to achieve motion, however, the amount of material fatigue
 474 is reduced by replacing some of the elastic hinges with pin joints. The
 475 displacement inverters achieved a geometric advantage of 0.87, and the force
 476 inverters achieved a mechanical advantage of 1.59. There are no theoretical
 477 values to compare with due to the elastic distortion required for actuation.

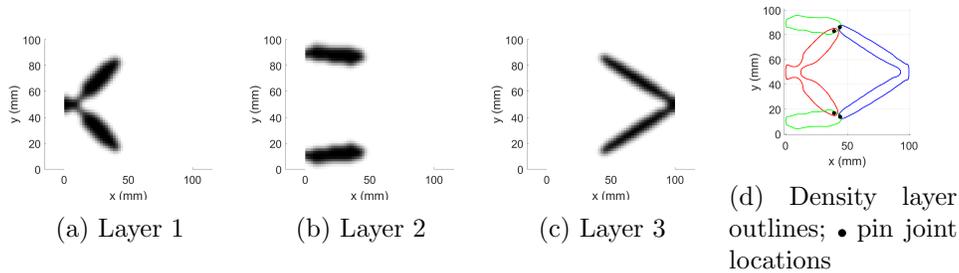


Figure 13: Optimal three-body force inverter

478 The inverter optimized for geometric advantage, pictured in Figure 12,
 479 has a layer (3) in which four elastic hinges form. The geometric advantage
 480 objective function does not take compliance into account, so designs that rely
 481 on elastic distortion to achieve large output displacements are not penalized.
 482 Large input forces may be necessary to create this motion, but the design's
 483 performance is not hindered according to the response functions.

484 Solutions were once again considered optimal once the KKT conditions
 485 had converged to a tolerance of 1×10^{-3} . Step limits of 2% of the parameter
 486 range were sufficient to reduce the KKT norm for the mechanical advantage
 487 inverter problem. For the inverter optimized for geometric advantage, the
 488 step limit was reduced by an order of magnitude to 0.2% of the parameter
 489 range to prevent oscillations and ensure convergence.

490 4. Conclusion

491 A method is proposed to optimize multi-body mechanisms via topol-
 492 ogy optimization. The added value over traditional topology optimization
 493 comes from the increased design space of multiple bodies connected with
 494 optimal connections. Pin joints are modeled by a Gaussian distribution of
 495 stiffness and allowed to migrate through the domain until an optimal com-
 496 bination of material distribution and pin joint location is determined. In
 497 other words, the connectivity is not defined a priori. Anchoring springs
 498 serve to adequately damp displacements to facilitate the nonlinear finite
 499 element solution.

500 Two-body grippers are optimized for both geometric and mechanical
 501 advantage. The theoretical mechanical advantage that is computed assum-
 502 ing rigid body rotation only is compared with the final mechanical advan-
 503 tage from the optimization. Numerical results indicate that there is strong

504 agreement between the two responses, therefore the damping and Gaussian
505 stiffness approximations do not unduly alter the design problem.

506 Force/displacement inverters are synthesized using a 3-layer design do-
507 main. Both results showed that multiple distinct bodies could form on the
508 same layer, which could significantly reduce the dimensionality of the opti-
509 mization problem as the number of layers are increased in more complicated
510 problems. The optimal designs for geometric and mechanical advantage both
511 use a combination of rigid body rotation and elastic distortion to achieve
512 optimal performance, although the amount of elastic distortion is reduced
513 compared with a single-body design.

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